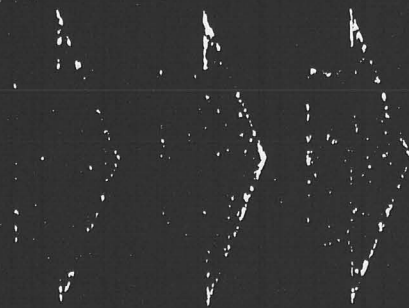




# *DISCOVERER RESEARCH*



## SUBSATELLITE

Declassified and Released by the N R C

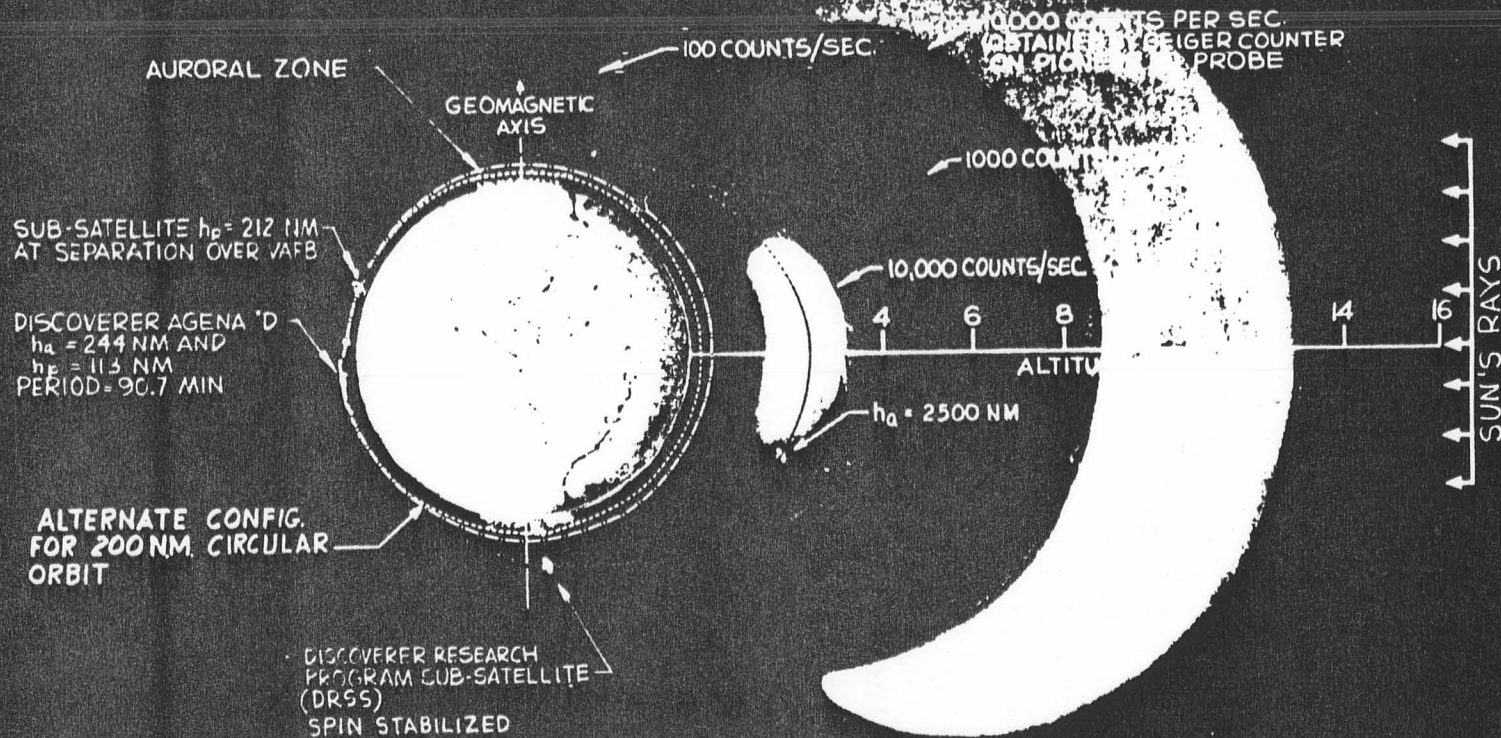
In Accordance with E. O. 12958

on NOV 26 1997

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# DISCOVERER RESEARCH PROGRAM

## SUBSATELLITE FLT CHARACTERISTICS



P-6176(1) 15C 1/30/62

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# DISCOVERER RESEARCH PROGRAM

## SUBSATELLITE

- TO PROVIDE A VEHICLE
  1. CAPABLE OF INCREASED ORBIT RANGES
  2. WITH EXTENDED ACTIVE LIFE (2 TO 6 MONTHS)
  3. TO CARRY A WIDE VARIETY OF RESEARCH EXPERIMENTS

- SUBSATELLITE CONFIGURED TO FIT ON AGENA D AFT RACK.
- THE SUBSATELLITE PROPULSION UNIT MAY BE VARIED TO ACHIEVE DIFFERENT ORBITS.
- BASIC STRUCTURE AND ELECTRONIC EQUIPMENT SUITABLE TO ACCOMMODATE RESEARCH EXPERIMENTS.
- ELECTRIC POWER MAY BE BATTERIES AND/OR SOLAR CELLS.

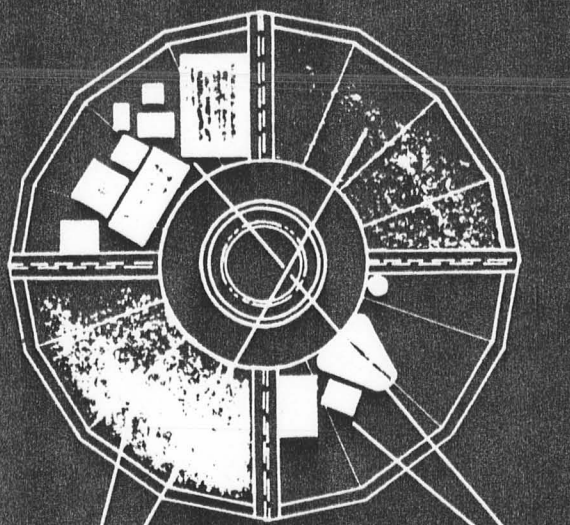
P-6946 "DISC" 4-30-62  
CY-5-1

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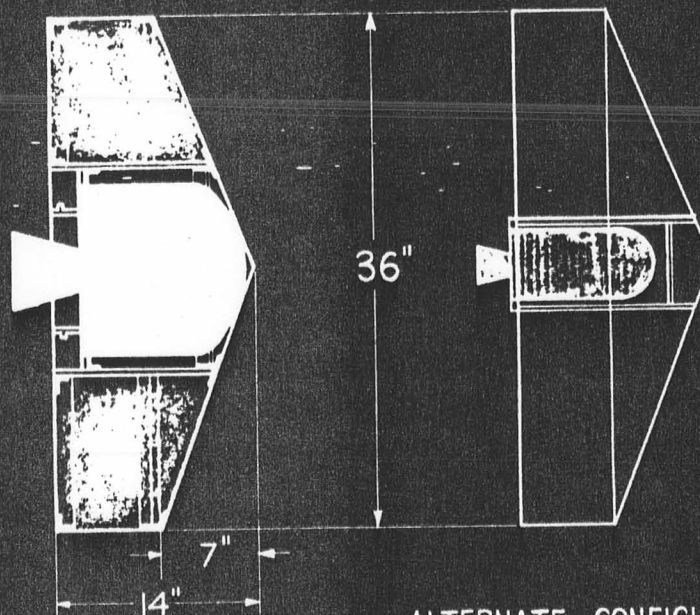
# DISCOVERER RESEARCH PROGRAM

## SUBSATELLITE DETAILS



BASIC EQUIPMENT

AVAILABLE SPACE FOR  
RESEARCH EXPERIMENTS



- TE 316 ROCKET
- 2000 NM APOGEE  
(FOR 225 LBS.)
- PAYLOAD SENSOR  
VOLUME = 3000 CU. IN.

- ALTERNATE CONFIGURATION
- TE 344 ROCKET
  - 200 NM CIRCULARIZED  
ORBIT
  - PAYLOAD SENSOR  
VOLUME = 3750 CU. IN.

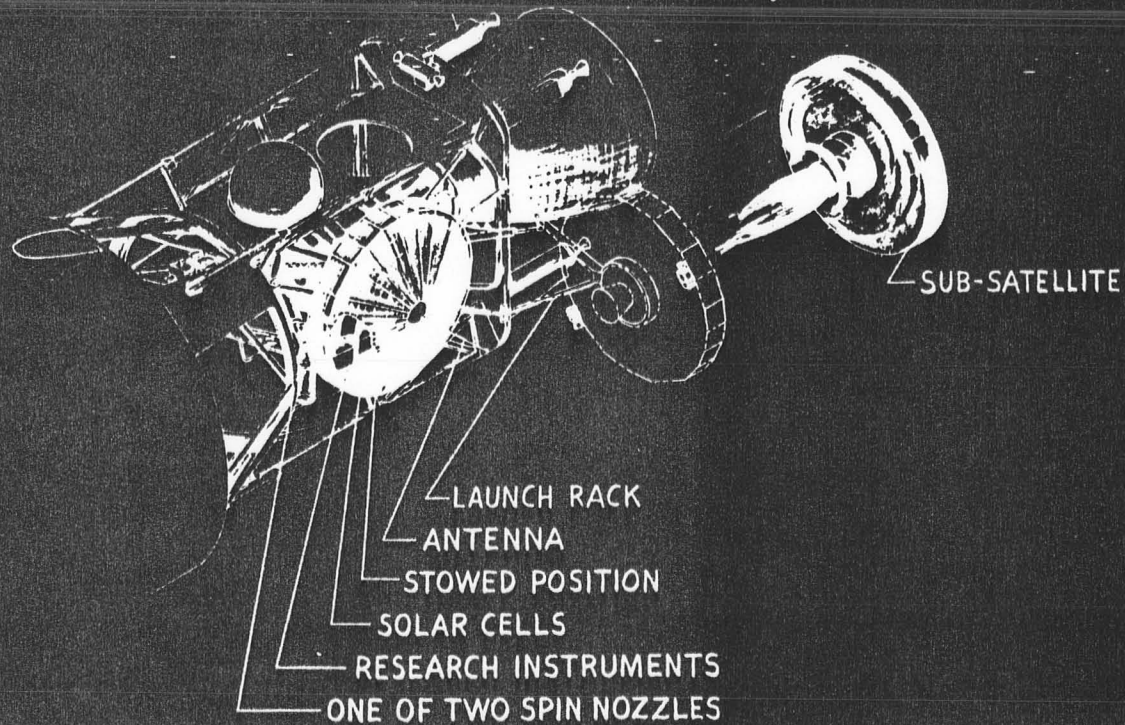
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# DISCOVERER RESEARCH PROGRAM

## LAUNCHING OF SUB-SATELLITE

AGENA D

DIRECTION OF FLT



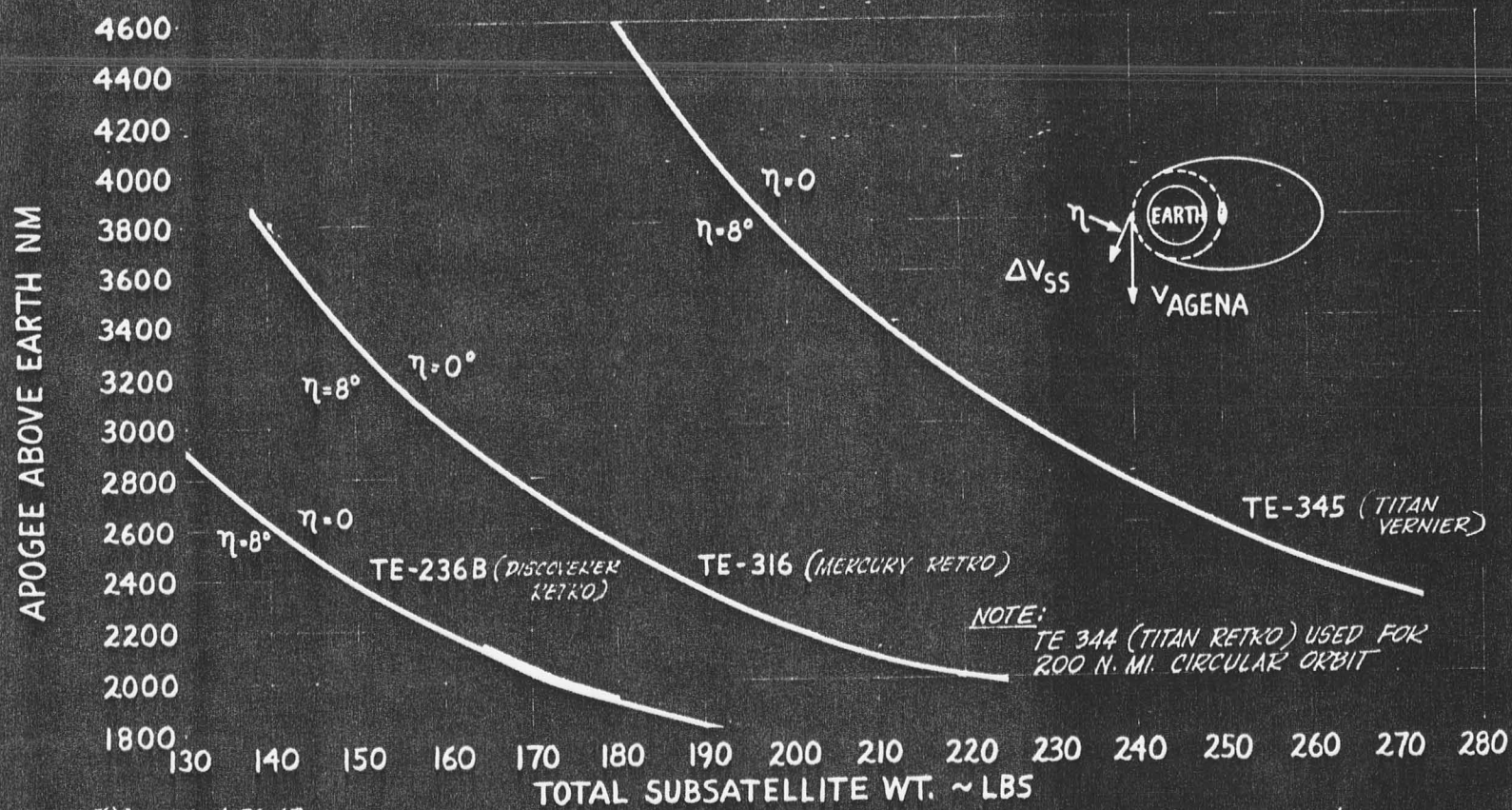
7-0595(1) DISC 4-30-62  
CY 5-1

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# DISCOVERER RESEARCH PROGRAM

## SUBSATELLITE WT. VS APOGEE



NOTE:  
TE 344 (TITAN RETRO) USED FOR  
200 N. MI. CIRCULAR ORBIT

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CV-5-1

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# DISCOVERER RESEARCH PROGRAM

## SUBSATELLITE EQUIPMENT & WEIGHT SUMMARY

DESCRIPTION	WEIGHT, LBS				
	TE 344	TE 236B	TE -316	TE -345	
BASIC VEHICLE					
STRUCTURE					12.0
SPIN SYSTEM					3.0
BATTERY & CHARGER					19.0
SOLAR CELL ASSEMBLY					5.0
ANTENNA & COUPLER					4.5
TIMER					1.2
COMMAND & SQUIB PROGRAMMERS					3.0
COMMUTATORS ( 2 REQUIRED )					3.2
TRANSMITTERS (2)					2.0
• MOD AMPLIFIER					.1
• VCO ( 4 REQUIRED)					1.2
• TRAY ASSEMBLY					.4
• DC-DC CONVERTER					1.0
COMMAND RECEIVER / DECODER					4.0
TAPE RECORDERS (2)					13.0
ACCELEROMETER					.6
CONNECTORS & WIRING					7.0
SUBTOTAL					78.2
BOOST ROCKET (INCL. HEATER BLANKET)	8.0	60.0	68.8	86.0	
EXPERIMENT PAYLOAD	127.8	75.8	67.0	49.8	
SUBSATELLITE					214.0
AGENA EQUIPMENT & STRUCTURE (LAUNCHER, UMBILICAL, BAROMETRIC SWITCH)					11.0
TOTAL SYSTEM WEIGHT					225.0

P-6944 Disc 4-30-62  
CY-9-1

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# DISCOVERER RESEARCH PROGRAM

## 2500 N.M. ~ SUBSATELLITE POWER SUMMARY

DESCRIPTION	POWER (WATT)	DUTY CYCLE* (PERCENT)	W - HR* ORBIT
COMMAND RECEIVER	0.35	100	0.81
TAPE RECORDER	4.0	100	9.20
T/M TRANSMITTER	18.0	10	4.14
DC-DC CONVERTER			
MOD AMPLIFIER			
VCO (4 UNITS) @ 30 MA	3.36	10	.77
COMMUTATOR (2 UNITS) @ 1.6 W	3.6	100	8.26
TIME REFERENCE GENERATOR	0.5	100	1.15
COMMAND & SQUIB PROGRAMMER	1.5	10	0.35
RESEARCH INST.	13.0	100	29.90
TIMER	1.5	10	.35
TOTAL	≈ 46		≈ 55

\* OPERATIONAL DUTY CYCLE BASED ON 2.3 HRS. PER ORBIT, AVERAGE POWER OF 15 WATTS FOR SOLAR CELLS, AND A 1.75 AMP-HR NI-CAD BATTERY IS AS FOLLOWS:

2.3 HRS/ORBIT - ON  
FOUR ORBITS - OFF

\* ORBIT - 137 MIN. = 2.3 HRS

2-6646(1) "DISC" 4-30-62  
CV-5-1

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# DISCOVERER RESEARCH PROGRAM

## SUB-SATELLITE COMMAND LINK CALCULATION

	GAINS	LOSSES
TRANSMISSION SYSTEM		
R.F. POWER 1000 WATTS	+ 30 db	
ANTENNA GAIN	+ 10 db	
RANGE 5000 N. M. (SLANT RANGE)		-154 db
RECEIVING SYSTEM		
RECEIVER SENSITIVITY	+ 132 db	
ANTENNA GAIN		- 3 db
FADE LOSS		- 3 db
POLARIZATION LOSS		- 3 db
TOTALS	+ 172 db	- 163 db

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# DISCOVERER RESEARCH PROGRAM

## SUB SATELLITE TELEMETRY LINK CALCULATION

	GAINS	LOSSES
TRANSMISSION SYSTEM:		
R.F. POWER 2 WATTS	+3 db	
ANTENNA GAIN		-3 db
RANGE 5000 NM (SLANT RANGE)		-157 db
RECEIVING SYSTEM:		
RECEIVER SENSITIVITY	+139 db	
ANTENNA GAIN TLM-18	+28 db	
FADE LOSS		-3 db
POLARIZATION LOSS		-3 db
<hr/>		
TOTALS	+170 db	-166 db

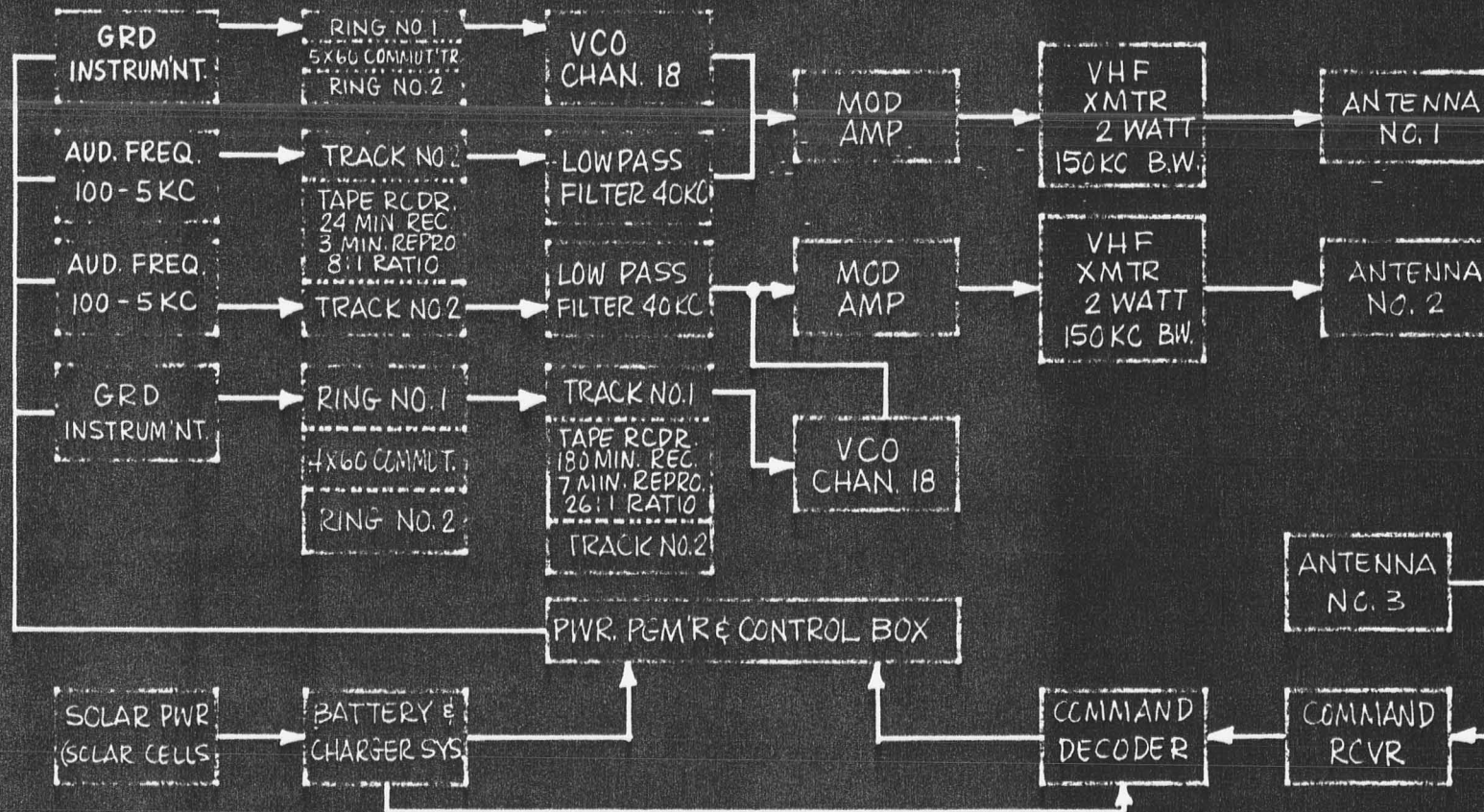
P-6638(1) "Disc" 4-30-62  
CW-3-1

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# DISCOVERER RESEARCH PROGRAM

## 200 N. MI ~ SUBSATELLITE TELEMETER SYSTEM



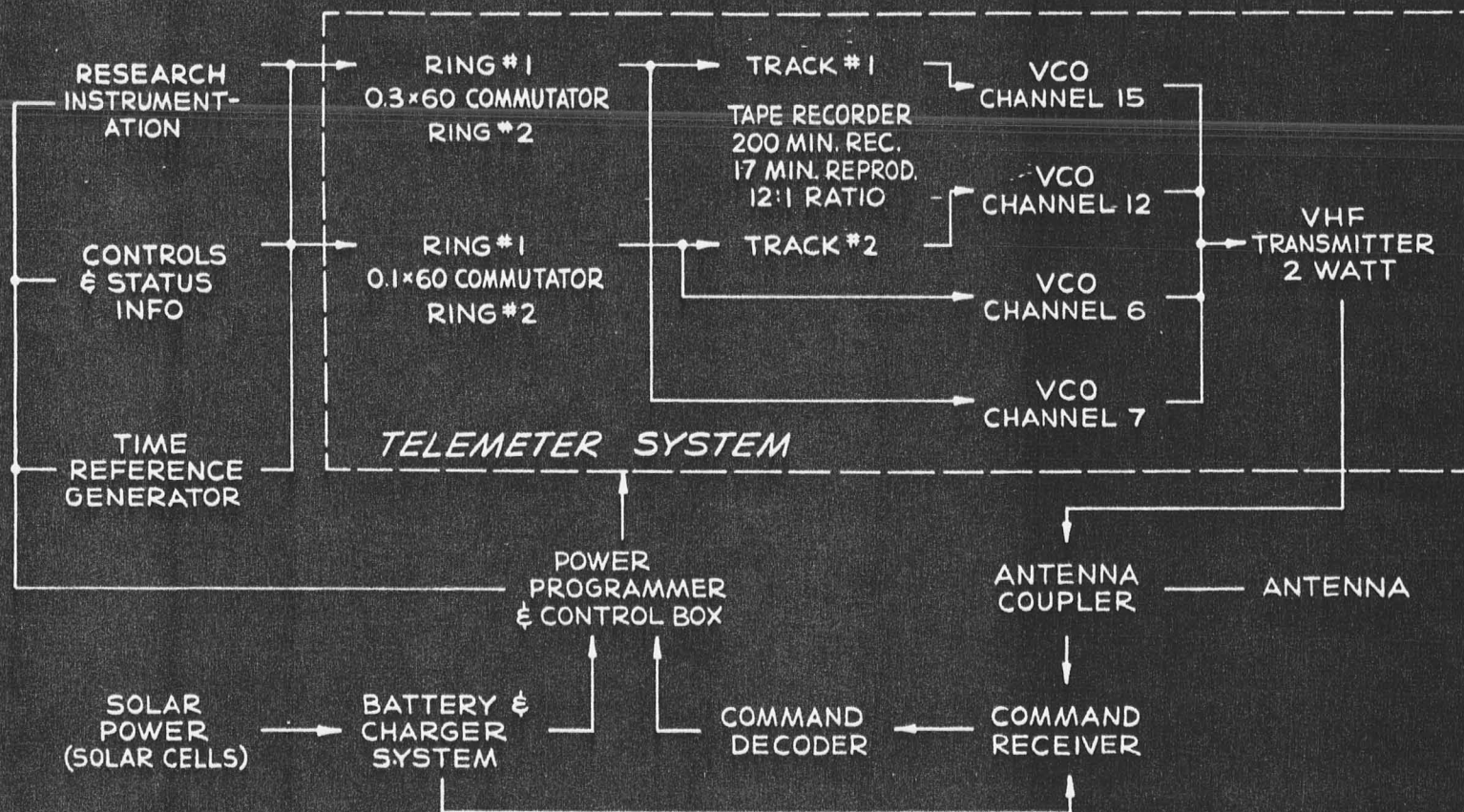
P-6985 DISC 5-1-62  
04-5-1

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# DISCOVERER RESEARCH PROGRAM

## 2500 N.M. SUBSATELLITE SYSTEM



P.6637(1) NSC 4-30-62  
C4-5-1

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# DISCOVERER RESEARCH PROGRAM

## SUBSATELLITE TRACKING

- ① "SPADATS" FOR PRIMARY EPHEMERIS DATA
  - ① DELETES NEED FOR TRACKING BEACON
  - ① MUCH MORE ACCURATE & RELIABLE THAN ANGLES ONLY
  
- ② SATELLITE TEST CENTER GENERATES ACQUISITION MESSAGES
  - ① ACQUISITION MESSAGES TO HAWAII & VANDENBERG, NTS TRACKING STATIONS
  
- ③ HAWAII & VANDENBERG FOR COMMAND & TELEMETRY DATA READOUT
  - ① NO ADDITIONAL MAJOR C & C EQUIPMENTS REQUIRED
  - ① NO ADDITIONAL COMMAND OR READOUT STATIONS REQUIRED
  
- ④ NEW BOSTON TRACKING STATION USED IN SUPPORT OF ABOVE FOR 200 N. MI. CIRCULAR ORBIT

P-6943 DISC 4-30-62  
CY-5-1

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# DISCOVERER RESEARCH PROGRAM

## SUBSATELLITE COST & PERFORMANCE COMPARISON

DESCRIPTION	BLUE SCOUT	DISCOVERER RESEARCH SUBSATELLITE
● COST		
○ BOOSTER & GUIDANCE	\$ 855,000	\$ 0
○ SATELLITE VEHICLE (INCL. PAYLOAD INTEGRATION, TELEMETRY, BATTERIES)	310,000	310,000
○ LAUNCH	GFE	0
○ TRACKING	GFE	GFE
○ TOTAL COST EACH FLIGHT	<u>\$ 1,165,000</u>	<u>\$ 310,000</u>
● PERFORMANCE (300 NM CIRCULAR ORBIT)		
○ INCLINATION ANGLE	70°-105°	70°-105°
○ ORBITING WEIGHT	160 LBS	225 LBS
○ VEHICLE ACTIVE LIFE	1 MONTH	2 - 6 MONTHS
● COST PER LB IN ORBIT	\$7,300	\$ 1,380

P-6972 "Disc" 4-30-62  
CV-5-1

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# DISCOVERER RESEARCH PROGRAM

## SUBSATELLITE SYSTEM RELIABILITY

ROCKET MOTOR & SPIN SYSTEM 0.990

LAUNCHER MECHANISM 0.999

COMMAND SYSTEM 0.971

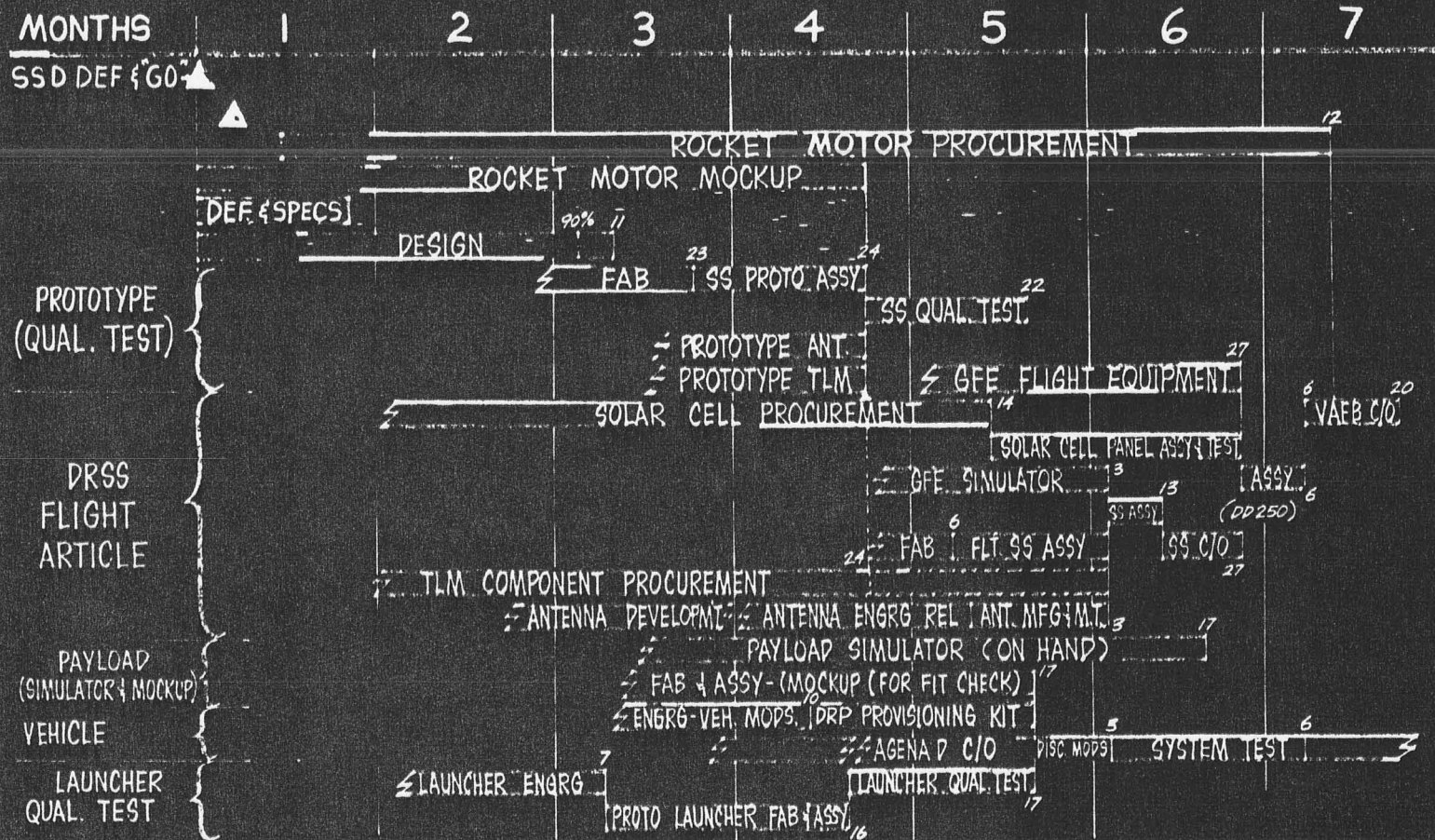
TELEMETRY SYSTEM 0.940

POWER SYSTEM 0.942

OVERALL SYSTEM 0.850

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# DRP SUBSATELLITE PROGRAM SCHEDULE



R. 6645(1) "DISC" 4-30-62  
CY-5-1

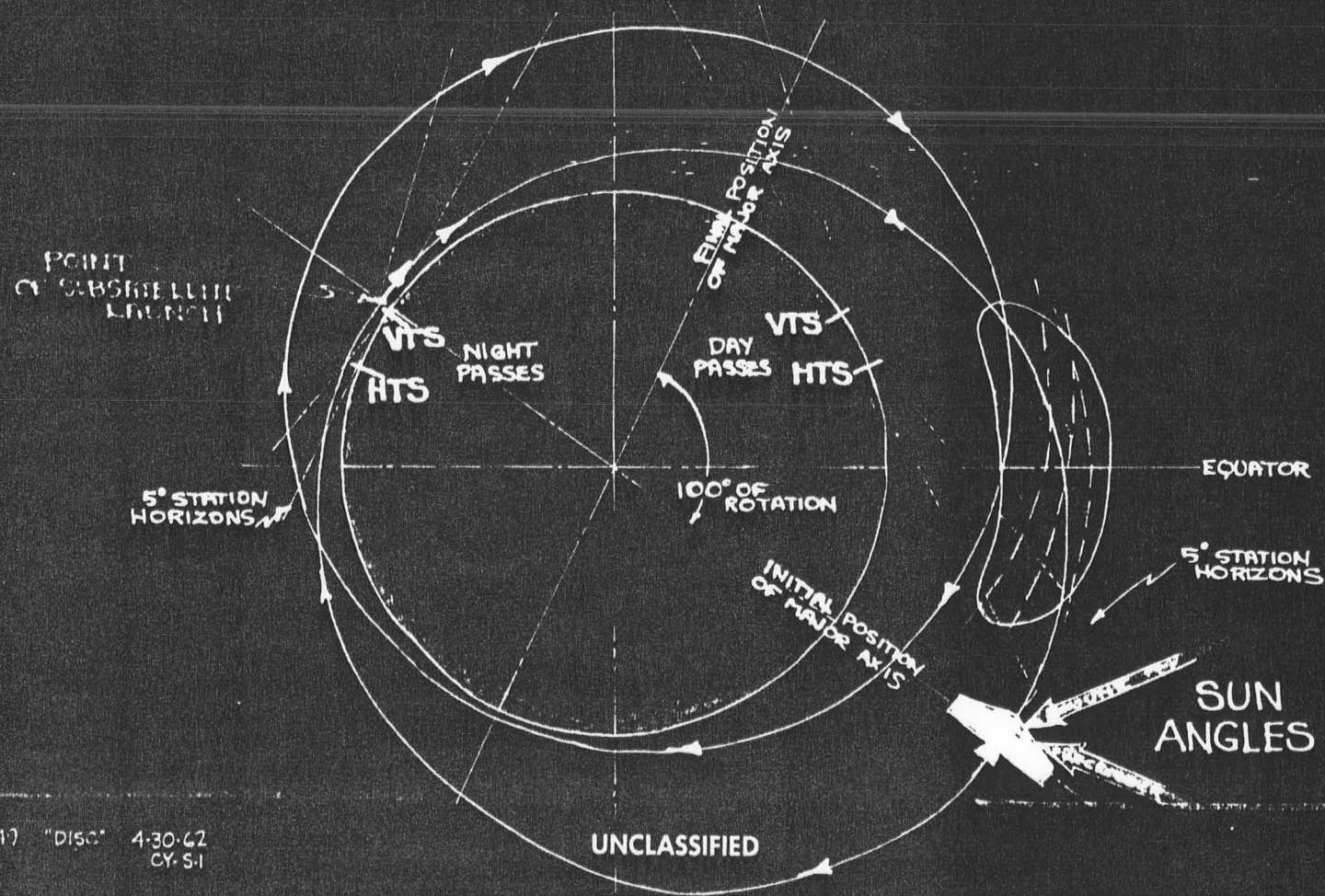
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# DISCOVERER RESEARCH PROGRAM

## SUBSATELLITE APOGEE ROTATION



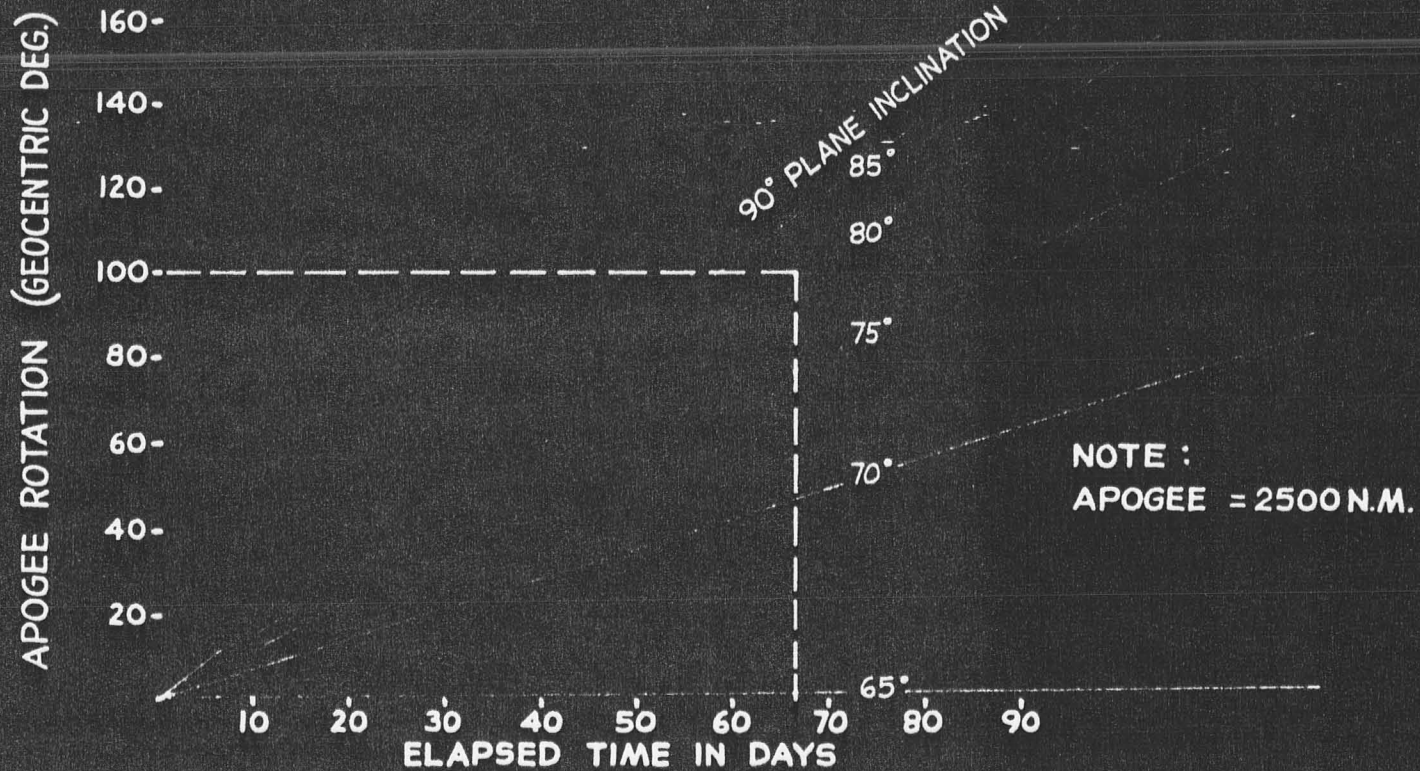
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CY-5-1

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# DISCOVERER RESEARCH PROGRAM

SUBSATELLITE APOGEE ROTATION VS ELAPSED TIME



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CY-S-1

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# DISCOVERER RESEARCH SUBSATELLITE

## SYSTEM RELIABILITY

DESCRIPTION	(4) DUTY CYCLE (PERCENT)	(1) LIFE (PERCENT PROBABILITY OF SURVIVAL)	(2) MEANTIME TO FAILURE (θ - HOUR)	COMPONENT RELIABILITY (R)
COMMAND SYSTEM				
RECEIVER	100	99.9 % FOR 100 HOURS	$10^5$	0.985 (3)
DECODER	100	99.9 % FOR 100 HOURS	$10^5$	0.985 (3)
			COMMAND SYSTEM RELIABILITY	0.971
TELEMETRY SYSTEM				
TRANSMITTER	10	MINIMUM OF 10,000 HOURS (95 % FOR 10,000 HOURS)	$2.0 \times 10^4$	0.999
DC-DC CONVERTER	10		$1.92 \times 10^5$	0.999
VCO	10	95 % FOR 2,000 HOURS	$3.84 \times 10^4$	0.999
COMMUTATOR (SWITCH TYPE)	100	95 % FOR 1,000 HOURS	$1.92 \times 10^4$	0.990
TAPE RECORDER	100	95 % FOR 200 HOURS	$3.84 \times 10^3$	0.954
TIME REF GEN.	100	98 % FOR 2,000 HOURS	$9.90 \times 10^4$	0.998
			TELEMETRY SYSTEM RELIABILITY	0.940
POWER SYSTEM				
SOLAR CELLS	100	95 % FOR 1440 HOURS (ESTIMATED)	$2.81 \times 10^4$	0.950
BATTERY	100	95 % FOR 115,000 HOURS	$2.25 \times 10^5$	0.994
POWER PROGRAMMER & CONTROL	100	95 % FOR 20,000 HOURS	$3.84 \times 10^5$	0.999
			POWER SYSTEM RELIABILITY	0.942

OVERALL SYSTEM RELIABILITY = 0.860

- NOTE: 1 PER INDIVIDUAL SPECIFICATIONS  
 2 ASSUME EXPONENTIAL DENSITY FUNCTION  $R = e^{-t/\theta}$   
 3 LIFE TEST ON SIMILAR EQUIPMENT CONDUCTED AT GEORGE C. MARSHALL SPACE FLIGHT CENTER - TOTAL NUMBER OF INTERROGATION CYCLES COMPLETED CORRESPONDS TO MAXIMUM INTERROGATION FOR 25 YEARS  
 4 ASSUMED - ACTIVE LIFE OF SUBSATELLITE OF 60 DAYS (3 HOURS OF OPERATION PER DAY)